



# X-37 Flight Demonstrator

*X-40A Flight Test Approach*



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# Phased Approach to Orbital Flight Demonstrations

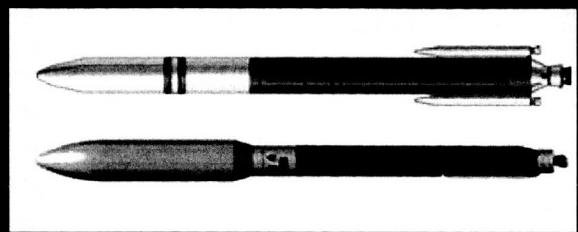
X-40A Completed Seven  
Successful Flights in 2001

Approach and Landing  
Test Vehicle Flies 2004

Orbital Vehicle  
Flies 2006 - 2007



On Orbit



EELV

B-52 will carry ALTV up to 40,000 feet



Drop Tests

Streamlined Ground Operations

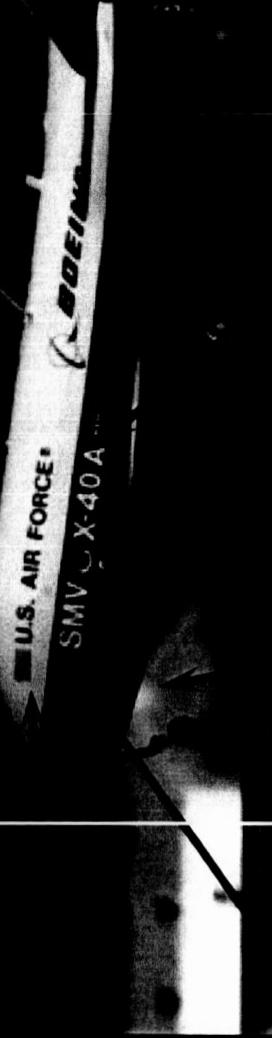
FLIGHT DEMONSTRATOR

# X-40A History

- Air Force owned vehicle developed by Boeing
  - Phase 1: One successful flight performed August 1998
  - Phase 2: Loaned to X-37 to provide early flight test data

## Improved Instrumentation & Telemetry

- Frequency agile
- Improved Data Rate
- Added Temp/Strain Sensors
- Improved Pressure Sensing
- Surface Position Sensors



Modification  
of surface  
actuator  
mounts

Modified Flight  
Termination System



Add X-37 Sensors  
• GPS/INS (SIGI)  
• Air Data (CADS)

Add Landing  
Gear Doors

FOCC Trailer Upgrades  
(Replaced Phase 1 DCV)

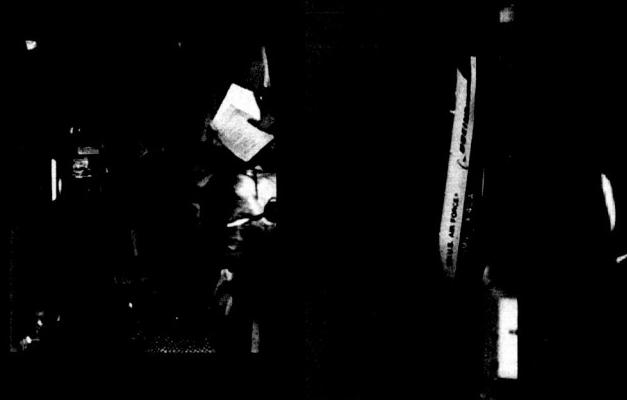
Upgraded Avionics Systems

- New dGPS Receiver/Modem and Command Receiver/Modem
- New uplink antennas
- Ruggedized Steering Controller & improved Actuator
- Addition of Heaters
- Power System Upgrades
- Additional Safing Systems

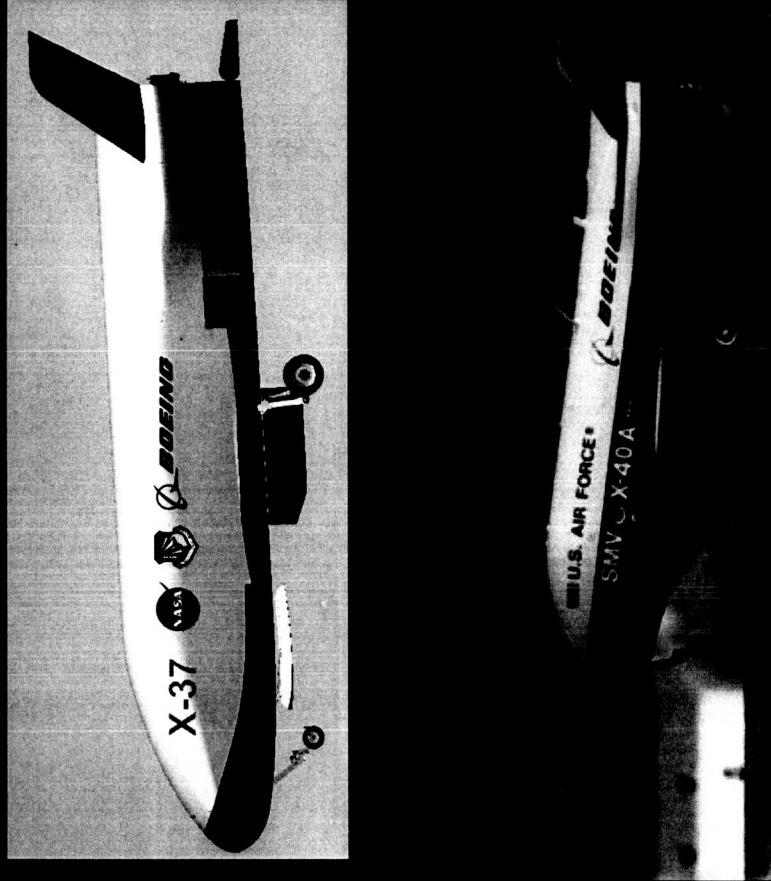
FLIGHT DEMONSTRATOR

# X-40A Flight Test Objectives

1. Evaluate Calculated Air Data System (CADS) experiment
2. Evaluate Honeywell SIGI (GPS/INS) under flight conditions
3. Flight Operation Control Center (FOCC) site integration and flight test operations
4. Flight test and tune GN&C algorithms
5. Conduct PID maneuvers to improve the X-37 aero database



# Comparison to X-37



## X-37

Landed Weight	7,500 lbs
Payload Weight	500 lbs
Fuselage Length	25.7 ft
Wing span	14.9 ft

## X-40A

Landed Weight	3,000 lb
Fuselage Length	21.5 ft
Wing span	11.5 ft
Length scale of X-37	80%

# Stepping Stone Flight Test Approach

Flight Tests Completed at DFRC with  
Cooperation from Edwards AFB



Free Flights



Captive Carry Flights



Tow/Taxi Tests

*Autonomous  
Approach & Landing*

# Tow/Taxi Test Objectives

- Verify navigation system performance
- Evaluate performance of X-37 navigation system experiment
- Verify closed-loop performance of the landing system
- Obtain data to substantiate landing system models
- Evaluate integrated X-40A subsystem operations

# Captive-Carry Flight Via CH-47 Helicopter

- Verify X-40A Vehicle & Ground System Performance
- Verify X-40A & Helicopter Performance and Behavior
- Rehearse GO/NO-GO and release procedures



# **Seven Successful Free-Fight Approach & Landing Tests From 15,000 Feet**

- Flight-to-flight conditions varied to gage vehicle response:
  - Released off centerline
  - Aerodynamic measurements made during vehicle maneuvers
  - Demonstrated unpowered flight and landing characteristics
- Collected data from onboard X-37 experiments
- Demonstrated operations concept

# The X-40A Flight Test Program Achieved Planned Objectives

- Develop Computed Air Data System (CADS) flight data to support X-37 system design
- Evaluation of Honeywell SIGI (GPS/INS) under flight conditions
- Flight Operation Control Center (FOCC) site integration and flight test operations
- Flight test and tune GN&C algorithms
- Base pressure instrumentation and PID maneuvers to improve the X-37 aero database

For More Information: